

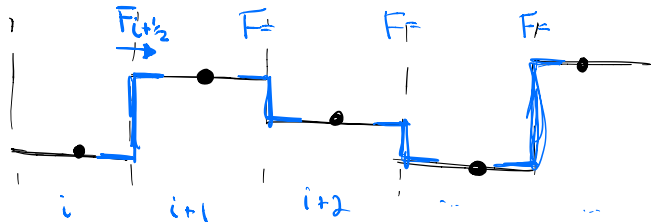
Recaps of stuff at end of Lecture 1

~ finite volume - like

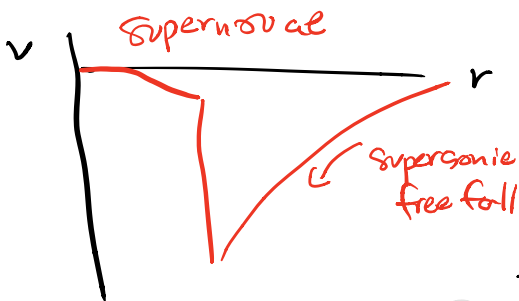
$$\frac{\partial \vec{U}}{\partial t} + \frac{\partial \vec{F}}{\partial x_j} = \vec{S} \Rightarrow \frac{U^{(n+1)} - U^{(n)}}{\Delta t} = S^{(n)} - \frac{F_{i+1/2}^{(n)} - F_{i-1/2}^{(n)}}{\Delta x}$$

$$\vec{U} = \begin{Bmatrix} \rho \\ \rho v_i \\ E \end{Bmatrix} \quad \vec{F} = \begin{Bmatrix} \rho v_j \\ \rho v_i v_j + P \delta_{ij} \\ (E + P) v_j \end{Bmatrix} \quad \vec{S} = \begin{Bmatrix} 0 \\ \rho a_i \\ \rho a_i v_j \end{Bmatrix}$$

$$U^{(n+1)} = U^{(n)} + \Delta t S^{(n)} - \Delta t \left( \frac{F_{i+1/2}^{(n)} - F_{i-1/2}^{(n)}}{\Delta x} \right)$$



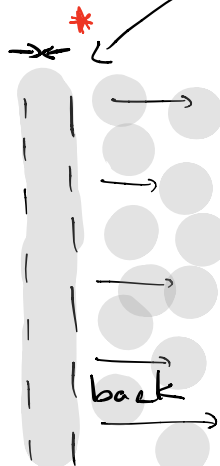
Some info on Shocks.



Astrophysically shocks are important too!

viscosity is crucial to converting KE to heat

$u_x^{(1)}, \rho^{(1)}, P^{(1)}$   
cold, upstream, ahead of shock  
front



$u_x^{(2)}, \rho^{(2)}, P^{(2)}$   
hotter, shocked region downstream behind

treat variable as discontinuous at shock.

Shock  $\rightarrow x$

mass is conserved across shock, mass does not accumulate in the shock

$$\rho^{(1)} u_x^{(1)} = \rho^{(2)} u_x^{(2)} \rightarrow [p u_x] = 0$$

mass flux in = mass flux out

\* strictly speaking this is the normal component of the velocity only. Tangential components of velocity do not change

momentum flux conservation

$$c_s^2 = \frac{\Gamma P}{\rho}$$

$$[p u_x u_j + P \delta_{xj}] = 0$$

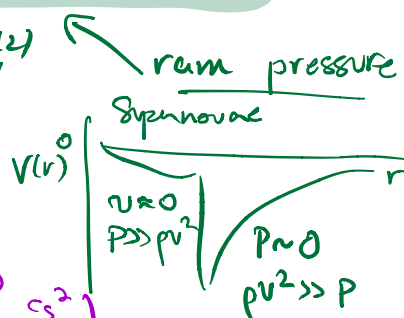
$$x=j \quad [p u_x^2 + P] = 0 \rightarrow \rho^{(1)} u_x^{(1)} u_x^{(1)} + P^{(1)} = \rho^{(2)} u_x^{(2)} u_x^{(2)} + P^{(2)}$$

$$x \neq j \quad \left. \begin{aligned} [p u_x u_y] &= 0 \\ [p u_x u_z] &= 0 \end{aligned} \right\} \rightarrow \rho^{(1)} u_x^{(1)} u_y^{(1)} = \rho^{(2)} u_x^{(2)} u_y^{(2)} \quad \leftarrow \text{ram pressure}$$

$$u_y^{(1)} = u_y^{(2)}$$

energy flux conservation

$$[u_x (p u_x^2 + p \varepsilon + P)] = 0 \Rightarrow \left[ \frac{1}{2} v^2 + h \right]_{(h = \frac{c_s^2}{\gamma-1})} = 0$$



→ if one has an EOS, these can be simplified  
ex if  $P = k \rho^\gamma$ , for strong shock ( $v_x^{(1)} \gg c_s^{(1)}$ )

$$c_s^2 = \frac{\gamma P}{\rho} \Rightarrow P = \frac{c_s^2 \rho}{\gamma}$$

ex. (1)

$$\frac{P_2}{P_1} = \frac{u_1}{u_2} = \frac{u_1^2}{u_2 u_1} = \frac{M_1^2 c_1^2}{u_2 u_1} = \frac{M_1^2 h_0 (\gamma-1)}{u_2 u_1 (1 + (\frac{\gamma-1}{2}) M_1^2)}$$

(3)

$$h_0 = \frac{1}{2} v_1^2 + \frac{c_{s1}^2}{\gamma-1}$$

$$c_{s1}^2 = \frac{h_0 (\gamma-1)}{(1 + (\frac{\gamma-1}{2}) M_1^2)}$$

$$\frac{(2)}{(1)} \Rightarrow \frac{P_1}{\rho_1 v_1} + v_1 = \frac{P_2}{\rho_2 v_2} + v_2$$

$$(v_1 - v_2) = \frac{1}{\gamma} \left[ \frac{c_2^2}{v_2} - \frac{c_1^2}{v_1} \right]$$

solve for

$$\frac{h_0}{v_1 v_2} = \frac{1}{2} \frac{(\gamma+1)}{(\gamma-1)}$$

$$\frac{M_1^2 (\gamma+1)}{2 + (\gamma-1) M_1^2}$$

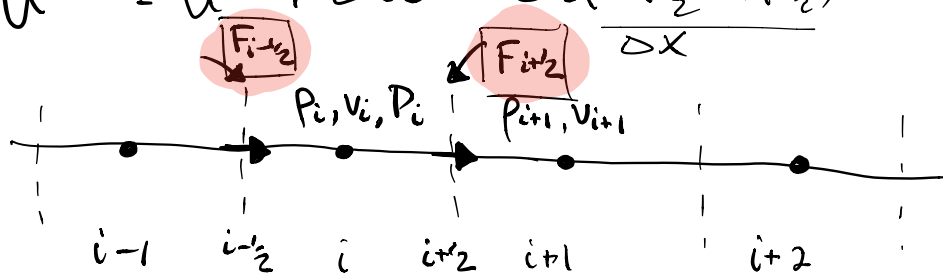
$$M_1 \gg 1$$

$$\frac{P_2}{P_1} = \frac{(\gamma+1)}{(\gamma-1)} \approx 7$$

$$\Gamma = 4/3$$

## Recall

$$U^{(n+1)} = U^{(n)} + \Delta t S^{(n)} - \Delta t \frac{(F_{i+\frac{1}{2}}^{(n)} - F_{i-\frac{1}{2}}^{(n)})}{\Delta x}$$



Steps to get  $U^{(n+1)}$ : Godunov method (for hydro, but also used in some rad traps)

0. determine timestep  $\Delta t$  (limited via CFL  $\Delta t \propto \frac{\Delta x}{c_s}$ )
1. Calculate  $U^{(n)}$  (prim2con)
2. Calculate sources  $S^{(n)}$  (sources)
3. Calculate Fluxes  $F_{i+\frac{1}{2}}^{(n)}$  &  $F_{i-\frac{1}{2}}^{(n)}$  (reconstruct & Riemann)
4. Determine  $U^{(n+1)}$  (evolve)
5. Determine new primitives (con2prim)

## Primitive vs. conservative

$$\vec{U} = \begin{Bmatrix} \rho \\ \rho v_i \\ \frac{\rho v_i^2}{2} + p\epsilon \end{Bmatrix}$$

$\vec{U}$  are called the conservative variables, but the fundamental variables are what we would like to have,  $\rho, v_i, \epsilon, P$ .

prim2con takes  $\rho, v_i, \epsilon, P \rightarrow \vec{U}$

con2prim takes  $\vec{U} \rightarrow \rho, v_i, \epsilon, P$

for these equations here these steps are trivial

if  $\vec{U} = (u_1, u_2^i, u_3)$

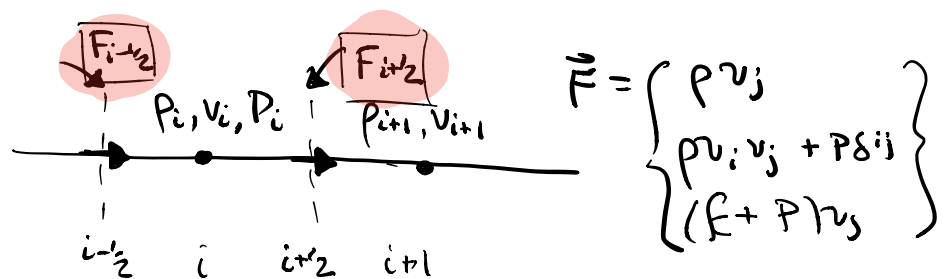
$\vec{Q} = (\rho, v_i, \epsilon, P)$

$\rho = u_1$        $\epsilon = \frac{u_3}{u_1} - \left(\frac{u_2^i}{u_1}\right)^2 \frac{1}{2}$

$v_i = u_2^i / u_1$        $P = \frac{2}{3} \epsilon$

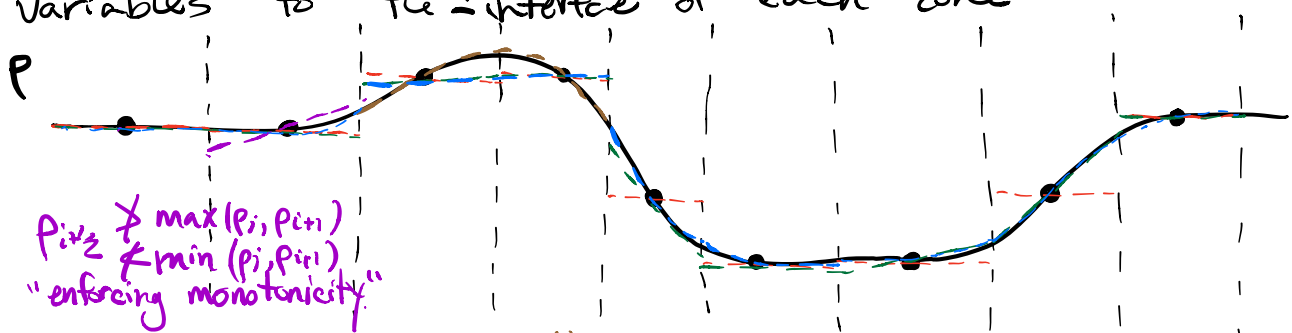
as we shall see, for relativistic hydrodynamics this is not the case.

# Fluxes



## Godunov method

1. Use some method to "reconstruct" cell centered variables to the  $\pm$  interface of each zone



Piecewise reconstruction  $0^{th}$

Piecewise linear + limiting (no new maximum)  $1^{st}$

Piecewise parabolic + limiting (PPM)  $2^{nd}$

→ lots of extra stuff

Flattening, steepening, fancy limiter

ENO/WENO methods

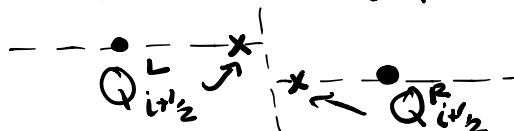
essentially non-oscillatory methods  
(W = weighted)  $3^{rd}$ ,  $5^{th}$ ,  $7^{th}$ ,  $9^{th}$

orders are in smooth flows

require cell  
averaging to  
stay the same

Colella & Woodward 1984

this method leaves you with two sets of reconstructed variables for each interface.



2. Solve the Riemann problem.

for HLL family of solvers  
Solvers determine  
 $F = R(F^L, U^L, F^R, U^R, \lambda)$

$$Q_{i+1/2}^L$$

L

$$Q_{i+1/2}^?$$

R

$$Q_{i+1/2}^R$$

$$Q_{i+1/2} = R(Q_{i+1/2}^L, Q_{i+1/2}^R)$$

How much flux is travelling across the boundary.

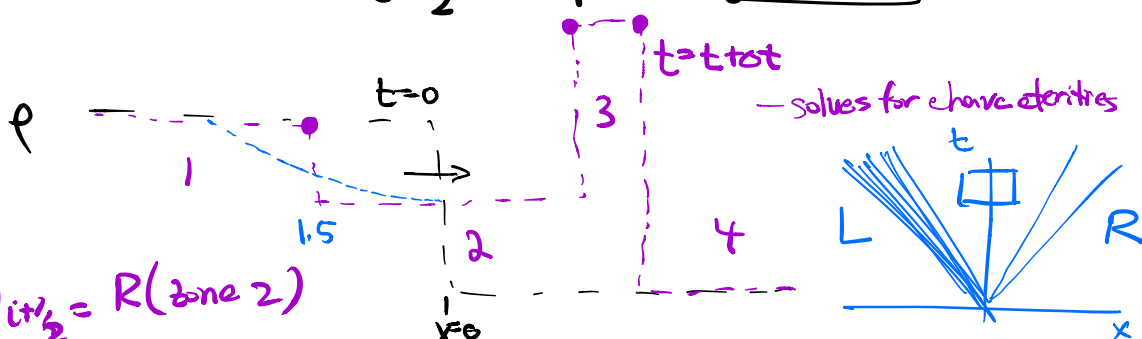
compute

$$F_{i+1/2} \text{ given } Q_{i+1/2}$$

need for step #3

$$i+1/2$$

Simple average is unstable



$$Q_{i+1/2} = R(\text{zone 2})$$

Open-source examples of Hydrocodes.

Pyro2, GIZM, FLASH, ET.  
ZEUS, ENZO, VHLI, Athena, Castro

$$F_{i+1/2} = \frac{\lambda_R F_L - \lambda_L F_R}{\lambda_R - \lambda_L} + \frac{\lambda_R \lambda_L}{\lambda_R - \lambda_L} (U_R - U_L)$$

Relativistic Hydrodynamics.

→ techniques are all the same, equations are different (but similar). Here, I'll assume the gravity is handled separately, therefore I'll take the metric as a given

$$g_{\mu\nu} : g = \det(g^{\mu\nu})$$

also assume we have a stress-energy tensor and a current density.

$$T^{\mu\nu} = \rho h u^\mu u^\nu + P g^{\mu\nu}$$

$$J^\mu = \rho u^\mu$$

$u$  = 4-velocity of the fluid  $u^\mu = (\gamma, \gamma v^i; \gamma v^i, \gamma u^k)$  flat, not moving  $u^i(1,0,0,0)$

$$h = (1 + \frac{\epsilon}{c^2} + P/\rho c^2)$$

$\epsilon$  = specific internal energy  
 $P$  = pressure

primitives:  $w = (\rho, v^i, \epsilon)$

Concept of conservation of energy, mass, and momentum fall out of

$$\nabla_\mu T^{\mu\nu} = 0 \quad \& \quad \nabla_\mu J^\mu = 0$$

$$\frac{\partial}{\partial x^\mu} (\sqrt{-g} J^\mu) = 0$$

Valencia formalism

define a set of conservative variables

$$D = \rho W$$

$$S_j = \rho h W^2 v_j$$

$$E = \rho h W^2 - P$$

$$\tau = E - D$$

$$W = \alpha u^0 = (1 - v^2)^{-1/2}; \quad v^2 = \gamma_{ij} v^i v^j$$

$$v^i = \frac{u^i}{\alpha u^0} + \frac{\beta^i}{\alpha}; \quad h = (1 + \frac{\epsilon}{c^2} + P/\rho c^2)$$

then expand  $\nabla \cdot ( ) = 0$  above to set:

$$\frac{1}{\sqrt{-g}} \left[ \frac{\partial}{\partial t} [\sqrt{-g} U] + \frac{\partial}{\partial x^i} [\sqrt{-g} F^i] \right] = S$$

$$\vec{U} = \begin{Bmatrix} D \\ S_i \\ \tau \end{Bmatrix} \quad F^i = \begin{Bmatrix} D(v^i - \beta^i/\alpha) \\ S_j(v^i - \beta^i/\alpha) + P \delta^i_j \\ \tau(v^i - \beta^i/\alpha) + P v^i \end{Bmatrix}$$

$$\vec{S} = \begin{pmatrix} 0 \\ T^{\mu\nu} \left( \frac{\partial g_{\nu j}}{\partial x^\mu} - \Gamma_{\nu\mu}^\delta g_{\delta j} \right) \\ \alpha \left( T^{\mu 0} \frac{\partial \ln(\alpha)}{\partial x^\mu} - T^{\mu\nu} \Gamma_{\nu\mu}^0 \right) \end{pmatrix}$$

\* Only complicated part compared to before is **con2prim**

the brentz factors & h complicate everything.  
con2prim requires a non-linear solve that should be iterated until convergence. added complication is the EOS needs to be invoked at each iteration.

$$\begin{aligned} D &= \rho W \\ S_j &= \rho h W^2 v_j \\ E &= \rho h W^2 - P \\ \tau &= E - D \end{aligned}$$

ex guess P

$$v_j^* = \frac{S_j}{\tau + D + P^*}$$

update  $W^*$

$$P^* = D / W^*$$

$$e^* = \frac{S_j}{\rho^* W^{*2} v_j^*} - P^* \rho^* - 1$$

Update EOS  $P^* = f(\rho^*, e^*)$   
iterate to converge on P.